

Heavy Ion Orbital Environment Single-Event Effects Estimations

ABSTRACT

This document discusses the methodology used to calculate on-orbit Single Event Effects (SEE) event rates.

Contents

1	Introduction	1
2	References	3

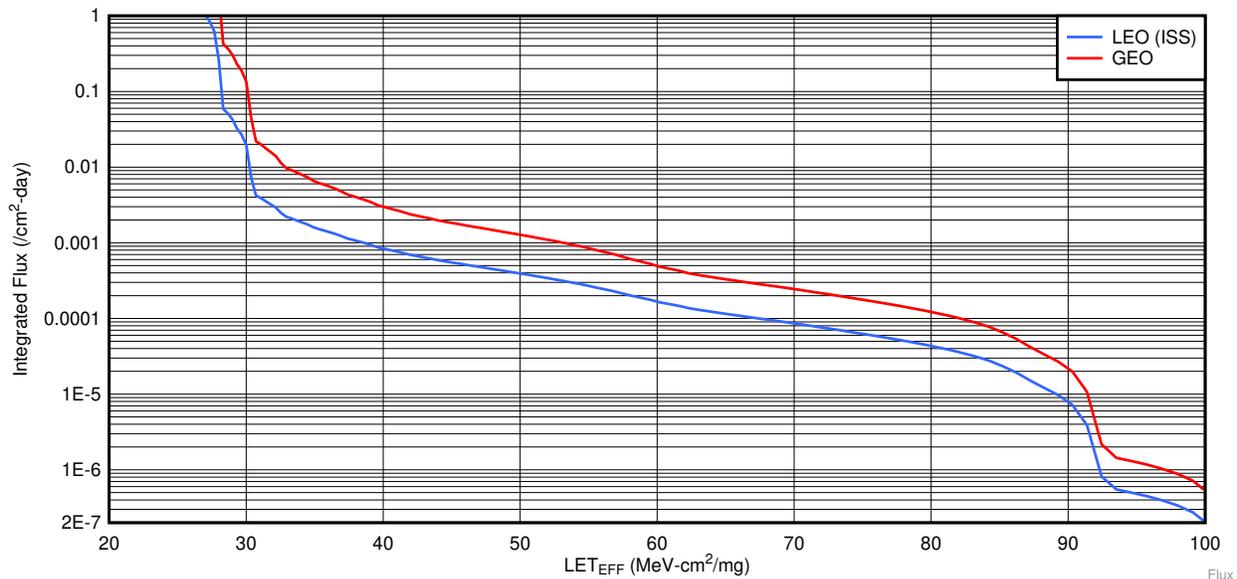
List of Figures

1	Integral Particle Flux Versus LET _{EFF} for a LEO-ISS (Blue Curve) and a GEO (Red Curve) Environment as Calculated by CREME96 Assuming Worst-week and 100 mils (2.54 mm) of Aluminum Shielding	2
2	Device Cross Section Versus LET _{EFF} Showing How the Weibull Fit (Green) is “Simplified” With the Use a Square Approximation (Red Dashed Line)	3

1 Introduction

To calculate SEE on-orbit event rates, both the device SEE cross-section and the flux of particles encountered in a particular orbit are required. Device SEE cross-sections are usually determined experimentally while flux of the particles in orbit is calculated using various software algorithms based on empirical data. For the purpose of generating representative event rates, a Low-Earth Orbit (LEO) and a Geostationary-Earth Orbit (GEO) were calculated using Cosmic Ray Effects on Micro-Electronics 96 (CREME96). CREME96 code is a suite of programs that enable estimation of the radiation environment in near-Earth orbits^[6, 4]. CREME96 is one of several tools available in the aerospace industry to provide accurate space environment calculations. Over the years since its introduction, the CREME models have been compared with on-orbit data and demonstrated their accuracy. In particular, CREME96 incorporates realistic “worst-case” solar particle event models where fluxes can increase by several orders-of-magnitude over short periods of time.

For the purposes of generating conservative event rates, the worst-week model (based on the biggest solar event lasting a week in the last 45 years) was selected. This event has been equated to a 99%-confidence level worst-case event^[6, 4]. The integrated flux includes protons to heavy ions from solar and galactic sources. A minimal shielding configuration is assumed at 100 mils (2.54 mm) of aluminum. Two orbital environments were estimated: that of the International Space Station (ISS), which is in LEO and the GEO environment. [Figure 1](#) shows the integrated flux (from high LET to low) for these two environments.



Note that the y-axis represents flux integrated from higher LET to lower LET. The value of integral flux at any specific LET value is actually the integral of all ion events at that specific LET value to all higher LETs.

Figure 1. Integral Particle Flux Versus LET_{EFF} for a LEO-ISS (Blue Curve) and a GEO (Red Curve) Environment as Calculated by CREME96 Assuming Worst-week and 100 mils (2.54 mm) of Aluminum Shielding

Figure 1 shows the Integral Particle Flux versus LET_{EFF} for an LEO-ISS (blue curve) and a GEO (red curve) environment as calculated by CREME96 assuming worst-week and 100 mils (2.54 mm) of aluminum shielding. Note that the y-axis represents flux integrated from higher LET to lower LET. The value of integral flux at any specific LET value is actually the integral of all ion events at that specific LET value to all higher LETs.

Using this data, you can extract integral particle fluxes for any arbitrary LET of interest. To simplify the calculation of event rates, assume that all cross-section curves are square, meaning that below the onset LET, the cross-section is identically zero while above the onset LET, the cross-section is uniformly equal to the saturation cross-section. Figure 2 illustrates the approximation with the green curve being the actual Weibull fit to the data with the “square” approximation shown as the red-dashed line. This allows you to calculate event rates with a single multiplication, the event rate becoming simply the product of the integral flux at the onset LET, and the saturation cross-section. Obviously, this leads to an over-estimation of the event rate since the area under the square approximation is larger than the actual cross-section curve, but for the purposes of calculating upper-bound event rate estimates, this modification avoids the need to do the integral over the flux and cross-section curves.

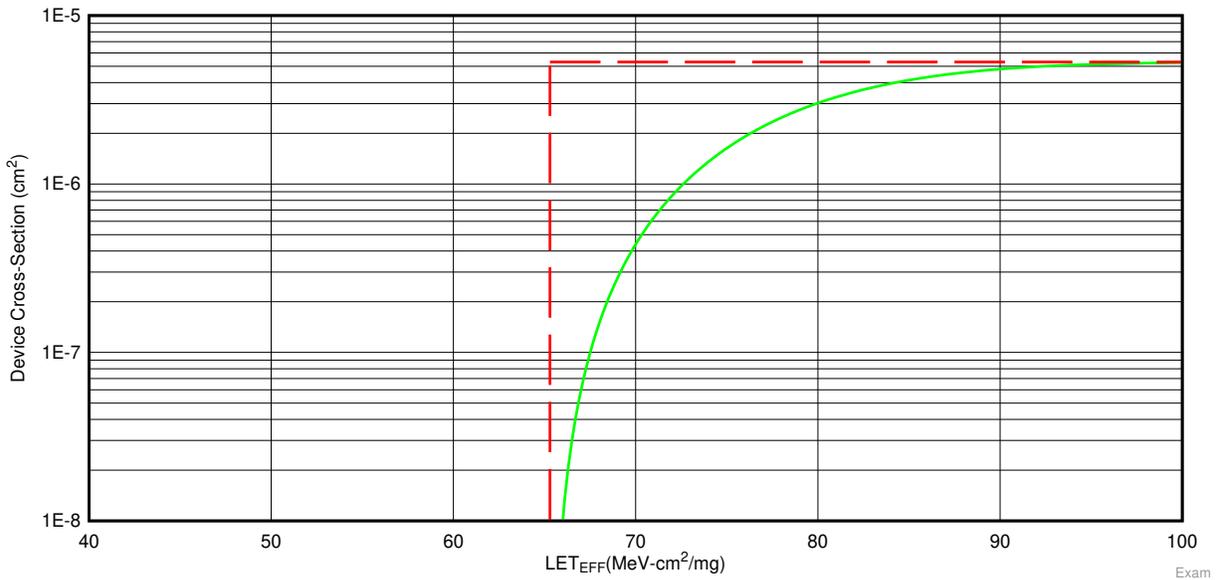


Figure 2. Device Cross Section Versus LET_{EFF} Showing How the Weibull Fit (Green) is “Simplified” With the Use a Square Approximation (Red Dashed Line)

Figure 2 shows a device cross section versus LET_{EFF}, showing how the Weibull fit (green) is “simplified” with the use a square approximation (red dashed line).

To demonstrate how the event rates are calculated, assume that you wish to calculate an event rate for a GEO orbit for the device whose cross-section is shown in Figure 2. Using the red curve in Figure 1 and the onset LET value obtained from Figure 2 (approximately 65 MeV-cm²/mg), you find the GEO integral flux to be approximately 3.24 × 10⁻⁴ ions/cm²-day. The event rate is the product of the integral flux and the saturation cross-section in Figure 2 (approximately 5.3 × 10⁻⁶ cm²):

$$GEO \text{ Event Rate} = \left(3.24 \times 10^{-4} \frac{\text{ions}}{\text{cm}^2 \times \text{day}} \right) \times (5.3 \times 10^{-6} \text{ cm}^2) = 1.71 \times 10^{-9} \frac{\text{events}}{\text{day}} \quad (1)$$

$$GEO \text{ Event Rate} = 0.71 \times 10^{-10} \frac{\text{events}}{\text{hr}} = 0.071 \text{ FIT} \quad (2)$$

$$MTBF = 1,607,820 \text{ Years} \quad (3)$$

2 References

1. <https://creme.isde.vanderbilt.edu/CREME-MC>
2. A. J. Tylka, and others, "CREME96: A Revision of the Cosmic Ray Effects on Micro-Electronics Code", IEEE Trans. Nucl. Sci., 44(6), 1997, pp. 2150-2160.
3. A. J. Tylka, W. F. Dietrich, and P. R. Boberg, "Probability distributions of high-energy solar-heavy-ion fluxes from IMP-8: 1973-1996", IEEE Trans. on Nucl. Sci., 44(6), Dec. 1997, pp. 2140 – 2149.
4. A. J. Tylka, J. H. Adams, P. R. Boberg, and others, "CREME96: A Revision of the Cosmic Ray Effects on Micro-Electronics Code", Trans. on Nucl. Sci., 44(6), Dec. 1997, pp. 2150 – 2160.

IMPORTANT NOTICE AND DISCLAIMER

TI PROVIDES TECHNICAL AND RELIABILITY DATA (INCLUDING DATASHEETS), DESIGN RESOURCES (INCLUDING REFERENCE DESIGNS), APPLICATION OR OTHER DESIGN ADVICE, WEB TOOLS, SAFETY INFORMATION, AND OTHER RESOURCES "AS IS" AND WITH ALL FAULTS, AND DISCLAIMS ALL WARRANTIES, EXPRESS AND IMPLIED, INCLUDING WITHOUT LIMITATION ANY IMPLIED WARRANTIES OF MERCHANTABILITY, FITNESS FOR A PARTICULAR PURPOSE OR NON-INFRINGEMENT OF THIRD PARTY INTELLECTUAL PROPERTY RIGHTS.

These resources are intended for skilled developers designing with TI products. You are solely responsible for (1) selecting the appropriate TI products for your application, (2) designing, validating and testing your application, and (3) ensuring your application meets applicable standards, and any other safety, security, or other requirements. These resources are subject to change without notice. TI grants you permission to use these resources only for development of an application that uses the TI products described in the resource. Other reproduction and display of these resources is prohibited. No license is granted to any other TI intellectual property right or to any third party intellectual property right. TI disclaims responsibility for, and you will fully indemnify TI and its representatives against, any claims, damages, costs, losses, and liabilities arising out of your use of these resources.

TI's products are provided subject to TI's Terms of Sale (www.ti.com/legal/termsofsale.html) or other applicable terms available either on ti.com or provided in conjunction with such TI products. TI's provision of these resources does not expand or otherwise alter TI's applicable warranties or warranty disclaimers for TI products.

Mailing Address: Texas Instruments, Post Office Box 655303, Dallas, Texas 75265
Copyright © 2020, Texas Instruments Incorporated